

## Problem Set 3

Due on Mon. 7<sup>th</sup> Oct. in class.

Fall 2013

## • Problem 1

For the causal LTI system described by the difference equation

$$y_k + \frac{1}{2} y_{k-1} = x_k, \quad (1)$$

- By first finding out the  $z$ -transform and then applying a suitable substitution for  $z$ , determine the frequency response  $H(e^{j\omega})$  for the system. Also find out the frequency response against unnormalized frequency if the sampling frequency  $f_s = 10$  Hz.
- Find out  $y_k$ , for  $x_k = \delta[k]$ , and determine its discrete-time Fourier transform (DTFT) using the formula

$$Y(e^{j\omega}) = \sum_{k=-\infty}^{+\infty} y_k e^{-j\omega k}. \quad (2)$$

- Compare the results of (a) and (b), and state the reason for any similarities or differences between  $H(e^{j\omega})$  and  $Y(e^{j\omega})$ .

## • Problem 2

- For the Euler Forward  $s$ -to- $z$  approximation  $z \simeq 1 + Ts$ , find out the range of values of  $s$  (if any) in terms of  $T$  for which
  - a stable pole on the real axis in  $s$ -domain becomes unstable in  $z$ -domain.
  - an unstable pole on the real axis in  $s$ -domain becomes stable in  $z$ -domain.
- Repeat (a) for the Euler Backward  $s$ -to- $z$  approximation  $z \simeq \frac{1}{1 - Ts}$ .
- For  $H(s) = \frac{s+1}{s^2-9}$ , transform poles and zeros to  $z$ -domain using each of Euler Forward, Euler Backward and Tustin's  $\left(z \simeq \frac{2+Ts}{2-Ts}\right)$  methods. Comment on the change in stability of the system for each case, while varying  $T$  from 0 to 1; for example, using  $T = 0.01, 0.2, 0.5, 0.8,$  and 1. Also suggest which approximation method best preserves the system's stability characteristics.

## • Problem 3

In the schematic shown in Fig. 1, assume that the mass of the spacecraft plus gas tank,  $m_1$ , is 2000 kg and the mass of the probe,  $m_2$ , is 1000 kg. A rotor will float inside the probe and will be forced to follow the probe with a capacitive forcing mechanism. The spring constant of the coupling  $k$  is  $4.2 \times 10^6$ . The viscous damping  $b$  is  $5.6 \times 10^3$ .

- Write the equation of motion for the system consisting of masses  $m_1$  and  $m_2$  using the inertial position variables,  $y_1$  and  $y_2$ .
- The actual disturbance  $u$  is a micrometeorite, and the resulting motion is very small. Therefore, rewrite your variables with scaled variables  $z_1 = 10^6 y_1$ ,  $z_2 = 10^6 y_2$ , and  $v = 1000u$ .
- Put the equations in state-variable form using the state  $\mathbf{x} = [z_1 \ \dot{z}_1 \ z_2 \ \dot{z}_2]^T$ , the output  $y = z_2$ , and the input an impulse  $u = 10^{-3}\delta(t)$  Ns on mass  $m_1$ .
- Using the numerical values, enter the equations of motion into MATLAB in the form

$$\dot{\mathbf{x}} = \mathbf{F}\mathbf{x} + \mathbf{G}v, \quad (3)$$

$$y = \mathbf{H}\mathbf{x} + Jv, \quad (4)$$

and define the MATLAB system: `sysGPB = ss(F,G,H,J)`. Plot the response of  $y$  caused by the impulse with the MATLAB command `impz(sysGPB)`. This is the signal the rotor must follow.

- (e) Use the MATLAB command  $p = \text{eig}(F)$  to find out the poles (or roots) of the system and roots  $z = \text{tzero}(F, G, H, J)$  to find out the zeros of the system.

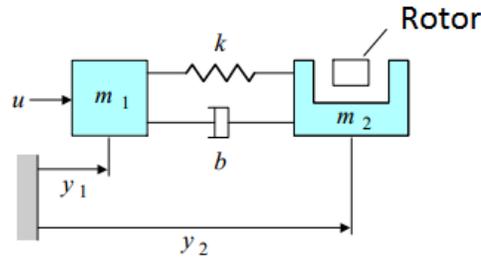


Figure 1: Mechanical set-up for Problem 3.

• **Problem 4**

Give the state description matrices in control-canonical form for the following transfer functions:

- (a)  $\frac{1}{3s + 2}$   
 (b)  $\frac{4s + 1}{s^2 + 5s + 4}$   
 (c)  $\frac{(s + 8)(s^2 + s + 25)}{s^2(s + 5)(s^2 + s + 36)}$

• **Problem 5**

For the system with state  $\mathbf{x}$  and described by the state matrices

$$\mathbf{F} = \begin{bmatrix} -2 & 1 \\ -2 & 0 \end{bmatrix}, \quad \mathbf{G} = \begin{bmatrix} 1 \\ 3 \end{bmatrix}, \quad \mathbf{H} = [1 \quad 0], \quad J = 0,$$

find out the transformation  $\mathbf{T}$  so that if  $\mathbf{x} = \mathbf{Tz}$ , the state matrices describing the dynamics of  $\mathbf{z}$  are in control canonical form. Compute the new matrices  $\mathbf{A}, \mathbf{B}, \mathbf{C}$ , and  $\mathbf{D}$ .

• **Problem 6**

For the system with state  $\mathbf{x}$  and described by the state matrices

$$\mathbf{F} = \begin{bmatrix} -4 & 1 \\ -2 & -1 \end{bmatrix}, \quad \mathbf{G} = \begin{bmatrix} 0 \\ 2 \end{bmatrix},$$

find out the steady-state value of the step-response assuming zero initial conditions.

• **Problem 7**

For the system shown in Fig. 2:

- (a) Find out the transfer function from  $U$  to  $Y$ .  
 (b) Write state equations for the system using the state variables indicated.

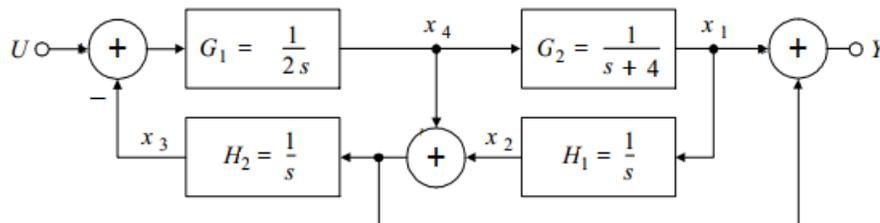


Figure 2: Block diagram for Problem 7.

• **Problem 8**

Consider the circuit in Fig. 3, with an input voltage source  $u(t)$  and an output current  $y(t)$ .

- (a) Using capacitor voltage and inductor current as state variables, write state and output equations of the system.
- (b) Find out the conditions relating  $R_1$ ,  $R_2$ ,  $C$ , and  $L$  that render the system uncontrollable.
- (c) Physically interpret the conditions found in (b) in terms of the time constants of the system.
- (d) Find out the transfer function of the system.

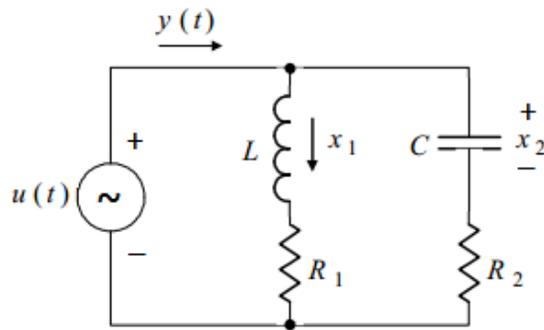


Figure 3: Circuit diagram for Problem 8.

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